

Report No.
NASA-CR-72075
Westinghouse
WAED 66.60E

DEVELOPMENT OF A 3400 WATT PROGRAMMED DC POWER SUPPLY PROVIDING STATIC AND DYNAMIC SOLAR CELL ARRAY SIMULATION

by

Paul Knauer

prepared for

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

Contract NAS 3-7920

2	167 - 3014	0
o Z	(ACCESSION NUMBER)	(THRU)
FOR	7 33	/
Ë	(PAGES)	(CÓDE)
LV CII	CR -72015	03
_	(NASA CR OR TMX OR AD NUMBER)	(CATEGORY)



WESTINGHOUSE ELECTRIC CORPORATION AEROSPACE ELECTRICAL DIVISION LIMA, OHIO

NOTICE

This report was prepared as an account of Government sponsored work. Neither the United States, nor the National Aeronautics and Space Administration (NASA), nor any person acting on behalf of NASA:

- A.) Makes any warranty or representation, expressed or implied, with respect to the accuracy, completeness, or usefulness of the information contained in this report, or that the use of any information, apparatus, method, or process disclosed in this report may not infringe privately owned rights; or
- B.) Assumes any liabilities with respect to the use of, or for damages resulting from the use of any information, apparatus, method or process disclosed in this report.

As used above, "person acting on behalf of NASA" includes any employee or contractor of NASA, or employee of such contractor, to the extent that such employee or contractor of NASA, or employee of such contractor prepares, disseminates, or provides access to, any information pursuant to his employment or contract with NASA, or his employment with such contractor.

Requests for copies of this report should be referred to

National Aeronautics and Space Administration Office of Scientific and Technical Information Attention: AFSS-A Washington, D.C. 20546

5

SUMMARY REPORT

Report No.

NASA-CR-72075

Westinghouse
WAED_66-60E

DEVELOPMENT OF A 3400 WATT PROGRAMMED DC POWER SUPPLY PROVIDING STATIC AND DYNAMIC SOLAR CELL ARRAY SIMULATION 4

by

Paul Knauer

prepared for

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

DE Contract NAS 3-7920 De la contract NAS 3-7920

Technical Management
NASA Lewis Research Center
Cleveland, Ohio
Bernard L. Sater

November 15, 1966

WESTINGHOUSE ELECTRIC CORPORATION

AEROSPACE ELECTRICAL DIVISION
LIMA, OHIO

DEVELOPMENT OF A 3400 WATT PROGRAMMED DC POWER SUPPLY PROVIDING STATIC AND DYNAMIC SOLAR CELL ARRAY SIMULATION

by

Paul Knauer

ABSTRACT

The primary objective in the design of the solar-cell array simulator was to provide a source of d-c power which would be similar, both statically and dynamically, to the output of a solar-cell array. To accomplish the latter, a large number of parallel power transistors operated as a Class-A series regulator were employed as the main controlling element. The solar-cell array volt-ampere characteristics were obtained by utilizing a diode function generator. The simulator operates in a closed-loop mode incorporating both voltage and current feedback. The resulting transient response obtained in the simulator is in the order of 50 microseconds.

PRECEDING PAGE BLANK NOT FILMED.

TABLE OF CONTENTS

<u>Title</u>	Page
SUMMARY	1
INTRODUCTION	2
BACKGROUND	2
THE SIMULATOR POWER STAGE	4
CONTROL PHILOSOPHY	5
THE FUNCTION GENERATOR	8
CLOSED-LOOP STABILITY	8
NOISE PROBLEMS	15
THE POWER-DISSIPATION LIMITER	16
ADDITIONAL PROTECTIVE FEATURES	18
MECHANICAL CONSIDERATIONS	18
POSSIBLE IMPROVEMENTS	19
CONCLUSIONS	23
APPENDIX - PARTS LIST	25

LIST OF ILLUSTRATIONS

Figure	<u>Title</u>	
1	Solar-Cell Simulator Output Characteristics	3
2	Schematic Diagram	6
3A	Typical Operational Amplifier Multiplier	7
3в	Simplified Schematic of Power Amplifier	7
4	Simplified Schematic - Function Generator	9
5	Transfer Characteristics of a Two-Diode Func- tion Generator	9
6	Frequency-Response Curves of Simulator	11
7A	Simulator transient response at 1-kHz load switching frequency. Load switches between 11 and 17 amperes	12
7B	Expansion of Figure 7A during load application.	12
7C	Expansion of Figure 7A during load removal	12
8A	Simulator transient response at 1 kHz load switching frequency. Load switches between 10 and 20 amperes	13
8B	Expansion of Figure 8A during load application.	13
8C	Expansion of Figure 8A during load removal	13
9A	Simulator transient response at 1-kHz load switching frequency. Load switches between 23 and 28 amperes	14
9B	Expansion of Figure 9A during load application.	14
9C	Expansion of Figure 9A during load removal	14
10	Loop Formed by Series Connection of Two Current Shunts	16
11A	Simulator output ripple under no-load conditions	17
11B	Simulator output ripple with 27 amp load applied	17

LIST OF ILLUSTRATIONS (Continued)

Figure	<u>Title</u>	Page
12	Front View of Simulator	20
13	Rear View of Simulator With Access Door Open	21
14	Top View of Simulator With Access Door Open	22

DEVELOPMENT OF A 3400 WATT PROGRAMMED DC POWER SUPPLY PROVIDING STATIC AND DYNAMIC SOLAR CELL ARRAY SIMULATION

by

Paul Knauer

Westinghouse Electric Corporation Lima, Ohio

SUMMARY

The purpose of this project was to develop a programmed d-c power supply capable of reproducing both the static and the dynamic characteristics of a solar-cell array. Simulation of the static characteristics was straightforward. The volt-ampere characteristics were determined by a diode-resistor function generator. Simulation of the dynamic characteristics, however, was one of the major design problems.

In order to obtain a transient response similar to that of a solar-cell array, a number of parallel power transistors were operated in a Class-A series regulator mode. This configuration eliminated the need for large filters normally associated with pulse-width-modulated control elements, and their inherent slow response. Consequently, the response time exhibited by the complete simulator was approximately 50 microseconds.

The quantity of parallel transistors was determined by the maximum power dissipation requirement of the supply, in this case 2500 watts. An efficient heatsink assembly utilizing forced-air cooling was used to minimize the number of transistors to 25.

Good engineering practice dictated that the simulator should be self-protecting. To this end, a power sensor was employed in a unique configuration which electrically maintained the power transistor dissipation below a safe value.

INTRODUCTION

The primary objectives of contract NAS3-7920 were twofold.

- 1. To develop power conditioning equipment to control and operate a Cesium Electron Bombardment Ion Thrustor System capable of operating from a solar-cell array energy source. (See NASA report CR-54672.)
- 2. To develop a power supply which would simulate, as closely as possible, the static and dynamic output characteristics of a solar-cell array, thereby allowing complete systems testing.

This report covers the second item only.

The completed power supply performs the intended functions well and, in addition, is self-protecting against static and transient load conditions of any kind. Illustrations which may be of immediate interest are:

Figure 1. Output Volt-Ampere Characteristics.

Figure 8. Photograph of Typical Transient Response

Waveforms.

Figure 12. Photograph of Complete Unit.

BACKGROUND

The static output characteristic range and capability which were to be simulated are shown in figure 1. From this figure, it was apparent that load currents exceeding 80 amperes could be expected. In order to control currents of this magnitude, a pulsewidth-modulated approach was considered from an efficiency standpoint. Pulse-width-modulated control may be obtained by using transistors, while thyristors would perform equally well. In either case, an appropriate L-C filter would be necessary to provide a smooth d-c output.

The closed-loop frequency response of a system utilizing pulse-width-modulation is inherently limited by the output filter, especially when load transients are such as to cause the regulator to attempt operation outside of its control limits, e.g., loo percent duty cycle. Attempts to raise the closed-loop response by increasing the regulator switching frequency would require switching in the range of loo-200 kHz. This was considered to be extremely difficult in view of the present state-of-the-art.

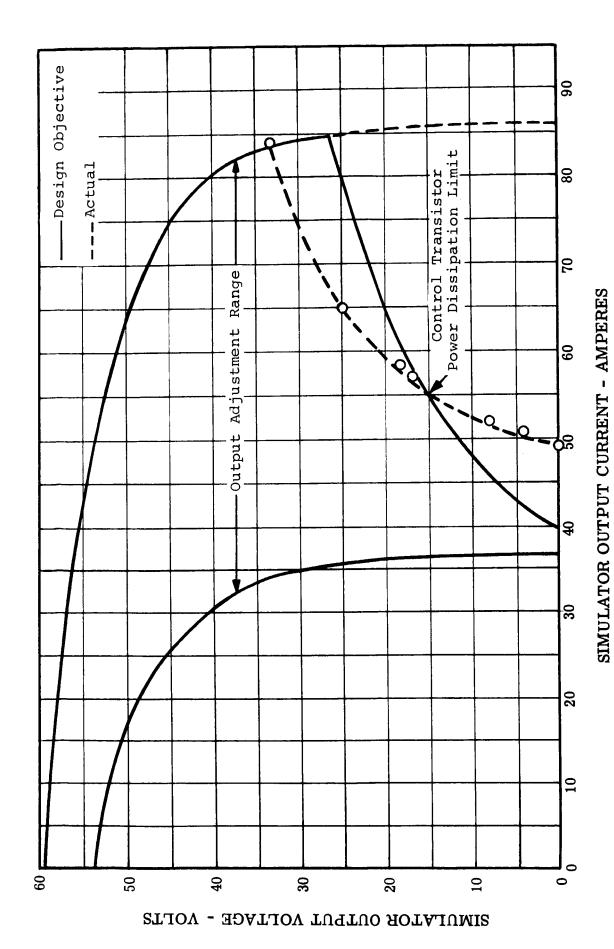


FIGURE 1. Solar Cell Simulator Output Characteristics

To overcome the frequency response limits imposed by an output filter, a power stage operated in a Class-A mode was investigated. The response time of a regulator of this type was limited by the transistors themselves, since an external filter was not required, and was therefore capable of considerably faster response. The main objection to this arrangement, of course, was the power dissipation requirement imposed upon the series control transistor. Referring to figure 1, dissipation in this transistor, under short-circuit conditions, is the product of the open-circuit voltage and the short-circuit current. In the design under consideration, the product exceeded 5000 watts.

A compromise between the pulse-modulated approach and the Class-A approach, a possible method of reducing this high dissipation, would be to incorporate thyristor control and Class-A transistor control in a series mode. In this manner, the transistors would allow rapid control over small load excursions, while the thyristor element with its associated filters would provide the coarse control required over large load excursions. Since this restricts the magnitude of the allowed load excursions, the Class-A series transistor approach was selected as the main controlling element.

THE SIMULATOR POWER STAGE

The simulator was expected to follow closely the solar-cell array characteristics during power conditioner tests, especially under short-circuit current conditions. At times, however, it may be desirable to operate two power conditioners from this supply, although no short-circuit tests would be conducted under these conditions. Both requirements are fulfilled by the simulator, as demonstrated by the characteristics curves shown in figure 1. In this approach, the power dissipation requirements of the series control transistors are held to a minimum by sensing and limiting the dissipation with a closed-loop control arrangement.

Thus simulator operation with only one power conditioner, which would require a short-circuit capability somewhat less than 40 amperes, would not enter the power-dissipation-limit region of the characteristics curve. The capability of operation with two power conditioners is also maintained past the maximum power point, where normal operation can be maintained.

The transistor power-dissipation capability utilizing this approach was 2500 watts. A parallel transistor approach was considered feasible only if the transistors could be forced to share load current. (This was ultimately accomplished by the addition of a low-value resistor into the emitter circuit of each of the paralleled transistors to provide sufficient current feedback to ensure current sharing.)

In order to employ as few transistors as possible, the transistors were selected on the basis of power dissipation capability. This required, in general, a transistor with a high current capacity, since a higher current requires a large semiconductor chip, which, by its very size, is capable of conducting heat to the transistor case more rapidly. Consequently, a Westinghouse type 2N2758 transistor was selected. Rated at 30 amperes, this transistor can safely dissipate 100 watts at a case temperature of 100° C. Under these conditions, the transistor junction temperature remains below 150° C, or 25° C below its rated tempera-Twenty-five parallel transistors of this type were required to form the main regulator controlling element. transistors are designated Q101 through Q125 in the schematic diagram shown in figure 2. Note that each transistor has an associated series fuse. Should any of the paralleled transistors fail shorted, that transistor would attempt to carry the entire load current. If this were to happen, the 5-amp fuse associated with the failed transistor would open, thereby removing the transistor from the line, and providing a visual indication of the failed transistor.

CONTROL PHILOSOPHY

Transistors Ql through Q4 provide the current amplification required to drive the main power transistors with the low-level signal available from operational amplifier Al.

Figure 3A shows a typical operational-amplifier circuit. It can easily be verified that, if the gain of the amplifier is sufficiently high and of the proper polarity, operation of the circuit can be described by the following equation:

$$E_{O} = \left(\frac{Rf}{Ri_{1}}\right) E_{1} + \left(\frac{Rf}{Ri_{2}}\right) E_{2}$$

In Figure 3B, a number of amplifier stages were added to the output of the amplifier A, but the equation still applies. In this circuit, however, the power output capabilities have been greatly increased.

This configuration, although somewhat simplified, is essentially derived from the schematic diagram, figure 2. Resistor R15 is the feedback resistor Rf, while R2 is input resistor Ri2. Any of the potentiometers R20 through R25 are equivalent to input resistor Ri1. If all the potentiometers just mentioned were removed from the circuit, then potentiometer R10 would determine

the value of E₁, while the ratio $\left(\frac{R16}{R2}\right)$ would determine the value

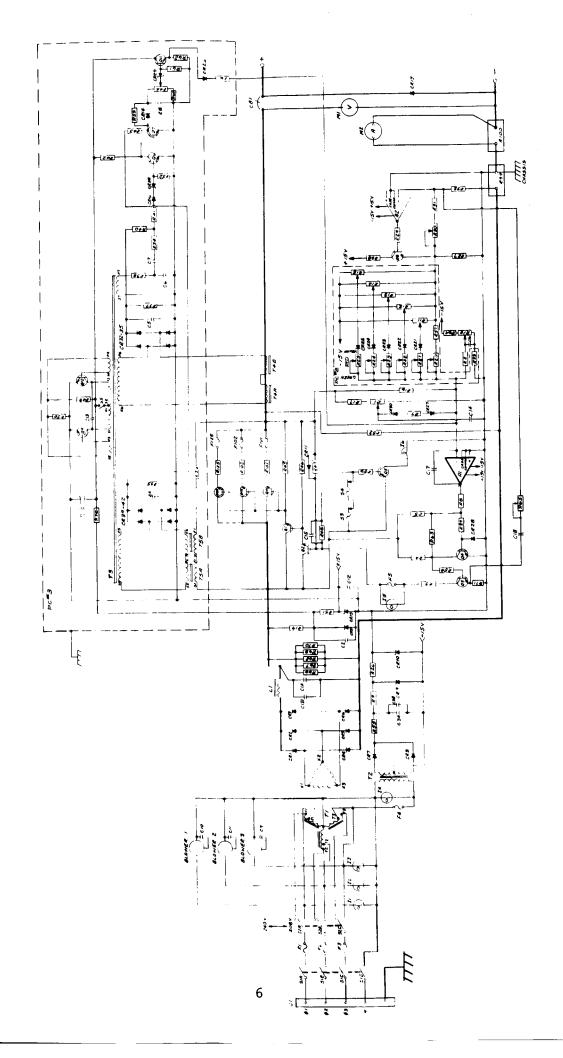


FIGURE 2. Schematic Diagram - Solar Cell Simulator

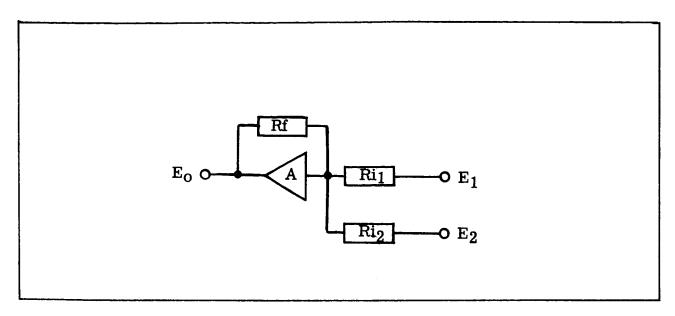


FIGURE 3A. Typical Operational Amplifier Multiplier

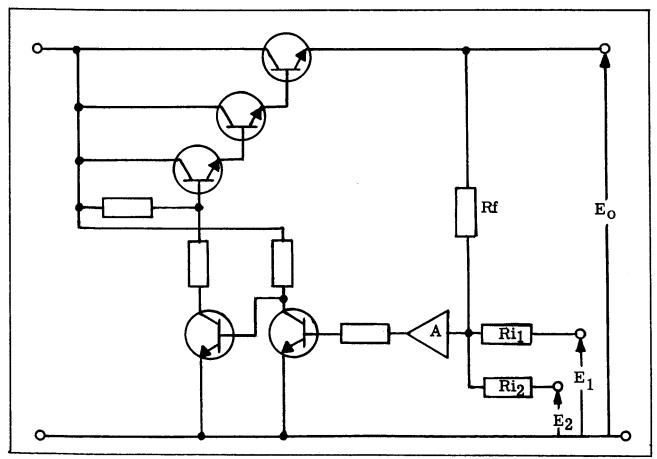


FIGURE 3B. Simplified Schematic of Power Amplifier

of $E_{\rm O}$. Under this condition, the simulator would be nothing more than a voltage regulator, with $E_{\rm O}$ remaining essentially constant regardless of the magnitude of the load current.

THE FUNCTION GENERATOR

Figure 4 represents a simplified schematic diagram of the simulator's function generator. The purpose of this circuit is to provide a control voltage which will ultimately shape the output characteristic of the simulator to match that of a solar-cell array.

The voltage E_O is merely a signal voltage proportional to the load current I_L. As long as the voltages at the taps of RA and RB are less than zero, the voltage E_G remains zero. If either of the voltages exceeds "zero", or the value of E_G, that voltage would add to E_G. Thus, the relationship between E_G and the load current I_L in this simplified circuit would approximate the curve shown in figure 5. It now becomes apparent that a proper choice of the number of breakpoints as well as the change in slope at each of the breakpoints can result in a fairly smooth curve of the desired shape.

The function generator used in the simulator is shown in figure 2 and consists of five breakpoints, determined by potentiometers R11 through R15. The desired slope associated with each of the breakpoints is determined by potentiometers R20 through R25. Any of these potentiometers may serve as the input resistor, Ri, shown in figure 3B.

In figure 2, potentiometer R30 serves to control the gain of the current-sensing circuit, thereby allowing the short-circuit current of the simulator to be adjustable.

An interesting phenomena is that the required forward breakdown voltage of each of diodes CR21 through CR25, normally considered to be a less than ideal characteristic, actually serves to round off the breakpoints of the resulting function generator output, such that the discontinuities are not discernable.

CLOSED LOOP STABILITY

The circuits described thus far form the functional part of the solar-array simulator, in that all additional circuits perform auxiliary functions such as rectification, filtering, and protection. These functions are described separately.

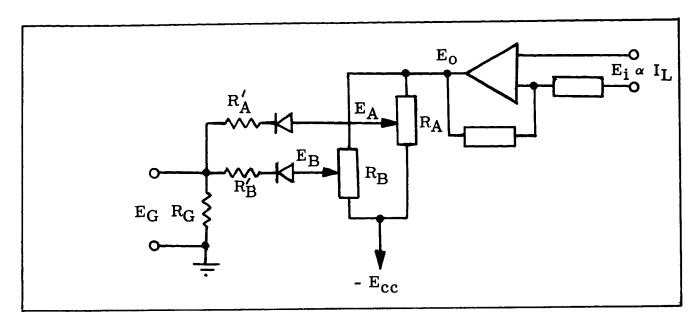


FIGURE 4. Simplified Schematic - Function Generator

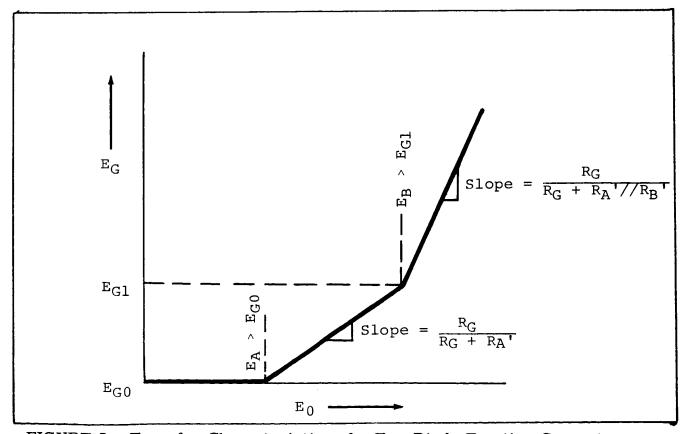


FIGURE 5. Transfer Characteristics of a Two Diode Function Generator

Stabilization of the closed loop was a greater problem than originally anticipated. An undesired closed loop was formed, for example, due to the capacitance between the various collector and base leads of transistors Q1, Q2, and the main power transistors. This loop caused oscillations within itself, which were eventually eliminated by reducing the frequency response of these transistors through the introduction of a capacitor across R48.

The main problem, however, was discovered after obtaining frequency-response curves of the simulator. These curves were obtained by disconnecting resistor R2 from R10 and capacitor C16, then applying a sinusoidal waveform of fixed magnitude to R2. The magnitude of the resulting waveform at the simulator output terminals was then measured to obtain the output/input voltage ratio at various frequencies.

The open-loop response curves were obtained without feedback resistor R16, while the closed-loop response curve was obtained with R16 in the circuit.

The original open-loop response curve, shown in figure 6, indicated four major breakpoint frequencies. The first, at 300 Hz, was the result of the operational amplifier Al, a Philbrick type P65AU. Both the second and the third breakpoints, at 1 kHz and 30 kHz, were due to transistor Q3, a Westinghouse type 2N3430. A fourth breakpoint existed in the region of 200 kHz, although its cause was not determined. Consequently, a gain decrease of 60 db per decade resulted at the 14 db crossover. This caused closed-loop oscillations of approximately 200 kHz.

The system was stabilized as follows. The Q3 transistor type 2N3430 was replaced with an MHT6310. This eliminated the breakpoints at 1 kHz and 30 kHz. Furthermore, a feedback capacitor was introduced across operational amplifier Al. This lowered the first breakpoint frequency. Thus, the operational amplifier was the only frequency limiting component above unity gain. This resulted in a stable closed loop. The resulting closed loop frequency response curve is shown in figure 6. It exhibits a roll-off frequency of about 20 kHz, which dictates the frequency response capabilities of the simulator. These capabilities are completely valid only for smaller load excursions, however, since the operational amplifiers are rate-limited devices. Consequently, the 50-microsecond response time indicated in figures 7, 8 and 9 under various load conditions implies a closed-loop roll-off frequency of only 5 kHz.

The basic criterion for stable closed-loop operation is to ensure that a 180° signal phase shift does not occur when the gain is greater than unity (Odb). This implies that the slope of the frequency response curve can not be greater than -20db/decade

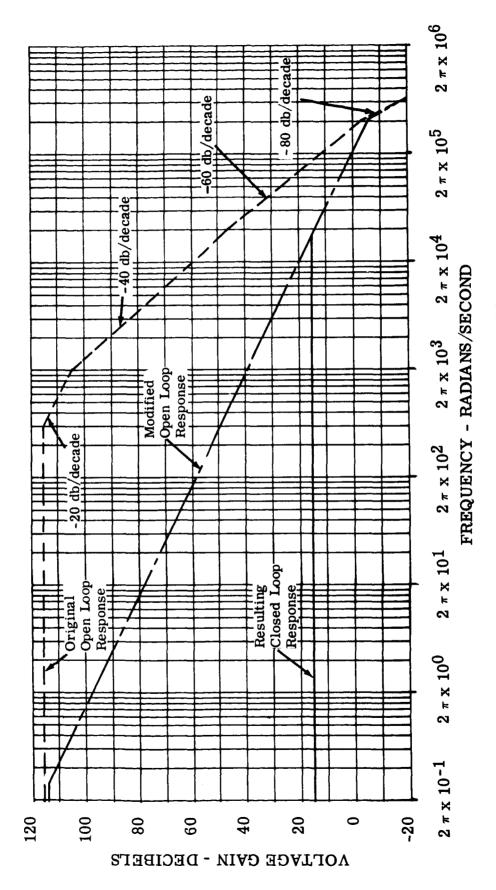


FIGURE 6. Frequency Response Curves of Simulator

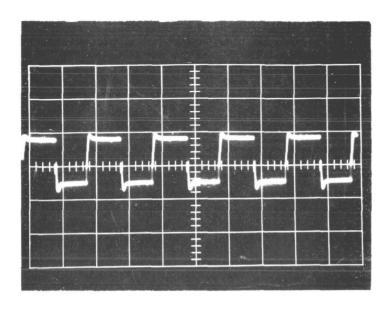


FIGURE 7A. Simulator transient response at 1 kHz loadswitching frequency. Load switches between 11 and 17 amperes

Horizontal: 0.5 msec/cm Vertical:

2.0 volts/cm

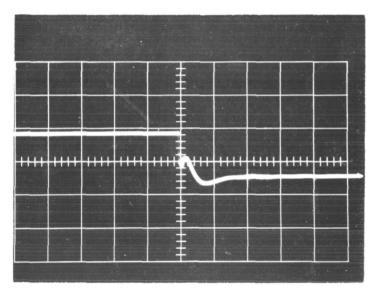


FIGURE 7B. Expansion of Figure 7A during load application

Vertical:

Horizontal: 50 µsec/cm 2.0 volts/cm

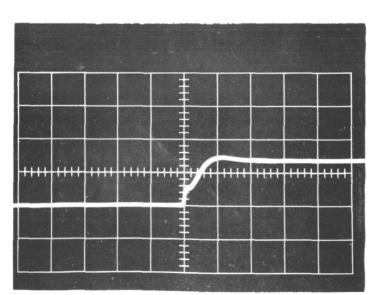


FIGURE 7C. Expansion of Figure 7A during load removal

Horizontal: Vertical:

 $50 \, \mu \text{sec/cm}$ 2.0 volts/cm

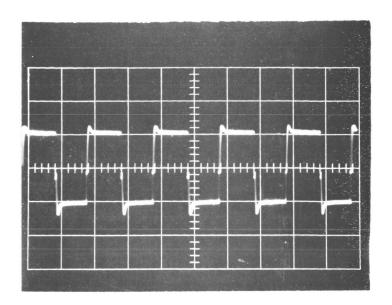


FIGURE 8A. Simulator transient response at 1 kHz load switching frequency. Load switches between 10 and 20 amperes

Horizontal: 0.5 msec/cm Vertical: 2.0 volts/cm

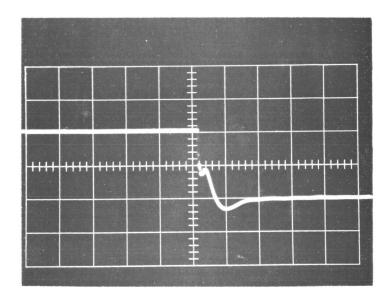


FIGURE 8B. Expansion of Figure 8A during load application

Horizontal: $50 \mu sec/cm$ Vertical: 2.0 volts/cm

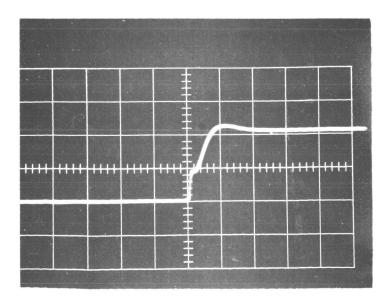


FIGURE 8C. Expansion of Figure 8A during load removal

Horizontal: $50 \mu \text{sec/cm}$ Vertical: 2.0 volts/cm

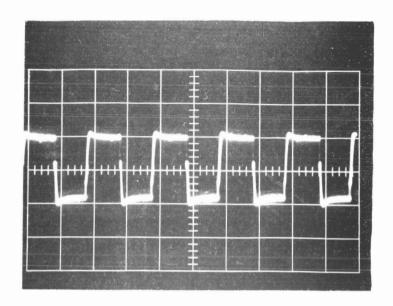


FIGURE 9A. Simulator transient response at 1 kHz load switching frequency. Load switches between 23 and 28 amperes

Horizontal: 0.5 msec/cm Vertical:

2.0 volts/cm

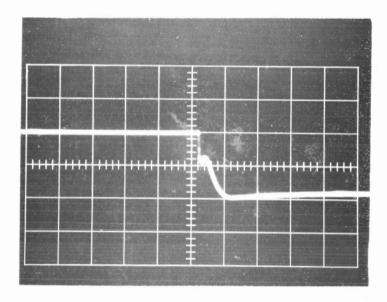


FIGURE 9B. Expansion of Figure 9A during load application

Vertical:

Horizontal: 50 µsec/cm 2.0 volts/cm

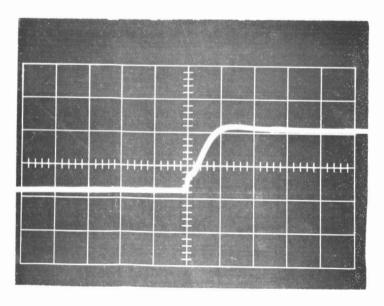


FIGURE 9C. Expansion of Figure 9A during load removal

Horizontal: Vertical:

 $50 \, \mu \text{sec/cm}$ 2.0 volts/cm above Odb. The simulator meets these requirements as shown by the closed-loop response curve in figure 6.

In order to increase the simulator response time, it would be necessary to shift the open-loop response curve of figure 6 to the right, while ensuring a -20db/decade slope above Odb. It is apparent that the breakpoint at 200 kHz is the limiting factor in any such attempt. Furthermore, the cause of this breakpoint was not clearly defined, but appeared to be attributable to such factors as stray wiring capacitance, transistor junction capacitance, and the like. A prelude to increased response capabilities must therefore be the analysis of these types of factors, and would no doubt result in requiring extreme care in the location of components and associated wiring during fabrication of a unit.

The resultant response of the simulator to changing load is illustrated photographically in figures 7, 8 and 9 under various conditions. The response time is typically 50 microseconds.

NOISE PROBLEMS

Electrical noise was a major concern in the design of the simulator. After stabilization, the d-c output was found to exhibit a considerable amount of undesired electrical noise. The prime sources of this noise were inductor Ll and the magnetic oscillator consisting of Q9, Q10, and associated components.

The inductor was located very close to current shunts R99 and R100. These shunts had been connected in series as shown in figure 10. Note that this connection formed a loop which was capable of producing a voltage when subjected to a changing electromagnetic field. This induced voltage was fed directly to operational amplifier A2. As a result of this injected signal, the ripple appearing at the output of the simulator was of such magnitude as to be unacceptable.

To reduce this effect as much as possible, the following action was taken. First, a non-inductive current shunt was obtained. This prevented the formation of a pickup loop since the output terminals were adjacent to each other. Second, the output of the shunt was increased by a factor of three in an attempt to increase the signal-to-noise ratio. The resulting configuration reduced the simulator noise output considerably. Figure 11 shows the resulting output ripple under various degrees of simulator loading.

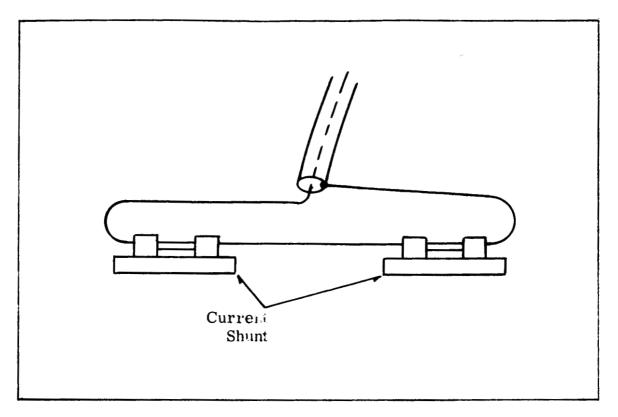


FIGURE 10. Loop Formed by Series Connection of Two Current Shunts

The noise injected into the system by the magnetic oscillator required two distinct, corrective steps. First, the entire oscillator circuit board was shielded with an aluminum cover to prevent radiated noise. Second, the oscillator was isolated from its power supply by means of an R-C filter in order to eliminate the high rate-of-change of current in the power-supply line. As a result, the peak noise caused by the oscillator was reduced to the extent that it did not exceed the peak noise already present at the output and shown in figure 11. Should it be desirable at any time to disable the oscillator, an internal switch has been provided for that purpose. Disabling the oscillator removes the protection feature of the power-limiting circuit to be described in another section. This feature, however, is only necessary when double output is required from the simulator.

THE POWER DISSIPATION LIMITER

As described previously in this report, a power-dissipation limiter was required in conjunction with the power transistors in

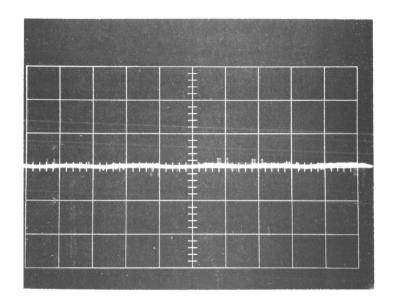


FIGURE 11A. Simulator output ripple under noload conditions

Horizontal: 5.0 msec/cm

0.5 volts/cm

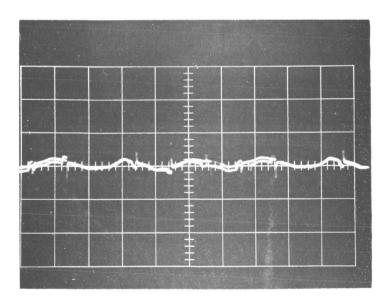


FIGURE 11B. Simulator output ripple with 27 amp load applied

Vertical:

Horizontal: 5.0 msec/cm 0.5 volts/cm

order to limit their number to twenty five. As evident from figure 1, however, the power limiting action is not necessary unless the simulator is required to deliver a short-circuit current exceeding 40 amperes. Referring to figure 2, the dissipation limiter senses the voltage across the power transistors by means of transformer T5, and the current through the power transistors by means of transformer T4. It utilizes a squarewave voltage source supplied by the magnetic oscillator in a configuration sometimes referred to as a transductor. Both transformers are merely saturable reactors which control the duration in which the square-wave is applied to the full-wave bridges. The duration is controlled in one case by the power-transistor voltage and in the other case by the power-transistor current. The resulting signals are rectified and filtered to produce voltages proportional to the variables being sensed. The two variables are then multiplied by allowing one to determine the pulse width, and the other the pulse amplitude of the collector voltage of transistor Q7. average of the resulting waveform is proportional to the power dissipation in the main transistors. When this voltage exceeds that of reference diode CR29, base drive to the main transistors is reduced, such that the resulting output characteristic resembles that shown in figure 1.

ADDITIONAL PROTECTIVE FEATURES

In addition to the power dissipation limiter, a number of safety features are incorporated in the simulator. Fuses protect both power transformers in the event of abnormal loading. A circuit breaker in series with the simulator output provides not only a convenient ON-OFF control, but protects the simulator in the event of a control circuit fault.

A free-wheeling diode, CR13, was incorporated to protect the simulator against any unusual transients which may occur during inductive load switching.

Likewise, diode CRII protects the main power and driver transistors by allowing a capacitive-load discharge through transistor Q3. Should this discharge current exceed safe values, fuse F5 will open. This, in turn, will energize I5, resulting in a front panel indication of the fuse failure.

Thermostatic switches S3 and S4 are located on each of the forcedair cooled heatsinks. A failure of either of the main blowers will cause one of the switches to open, thereby interrupting base drive to the main transistors. Simultaneously, lamp I6 will be energized to provide a front panel indication. The thermostatic switches reset automatically as soon as the temperature returns to a safe level.

MECHANICAL CONSIDERATIONS

Figure 12 shows a photograph of the simulator front panel. All controls were located for ease of access and readability, and were minimized for simplicity. Thus, the front panel contains a main power switch, an output circuit breaker, a control for open circuit voltage and a control for short circuit current.

Four green indicator lamps are energized when all input voltages are present, while two red indicator lamps are energized only under the abnormal conditions described previously.

Figures 13 and 14 show rear and top views of the simulator interior. The toggle switch located under the power receptacle in figure 13 selects one of two taps on the main power transformer to allow operation from 208-volt or 240-volt, 3-phase, 60-cycle power sources. An incorrect switch position is not harmful to the simulator, but may affect the output characteristics under certain load conditions. It is therefore important to maintain a check on the line-to-line voltage at periodic intervals.

The input connector includes a neutral connection and a chassis ground connection. The neutral connection may be connected, but is not necessary under normal conditions. Figure 14 shows a complete view of all fuses. Their location allows easy access for visual inspection and replacement.

POSSIBLE IMPROVEMENTS

Several areas where improvements in operation may be realized should be considered at this time. These are as follows:

- 1. A revised layout of the simulator components should acknowledge those components and circuits which produce electrical noise as well as those which are susceptible to electrical noise. This action would further reduce the undesirable signals injected into the system such as those shown in figure 11.
- 2. A gain increase in the power-dissipation limiter loop should be considered, in order to allow limiter operation closer to the constant-power limit curve shown as a design objective in figure 2.
- 3. A detailed analysis of the uppermost breakpoint frequency shown in figure 6 would determine the cause of



FIGURE 12. Front View of Simulator

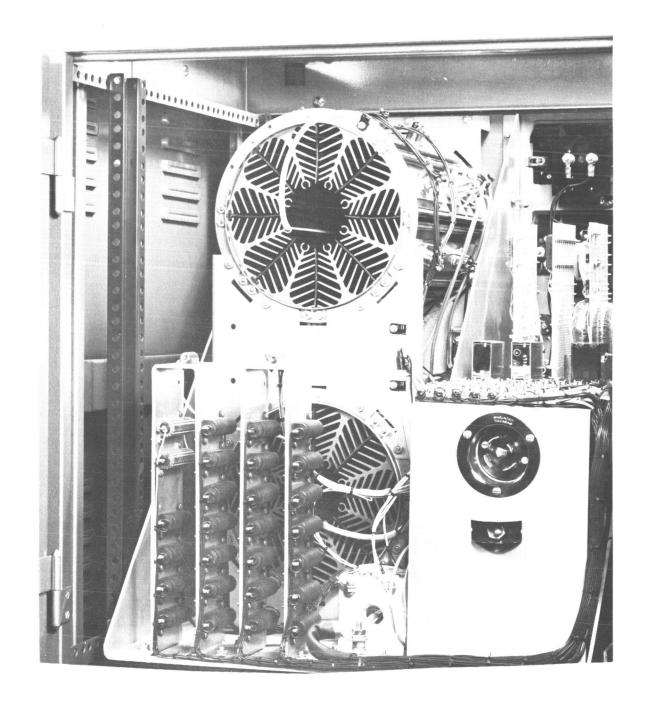


FIGURE 13. Rear View of Simulator With Access Door Open

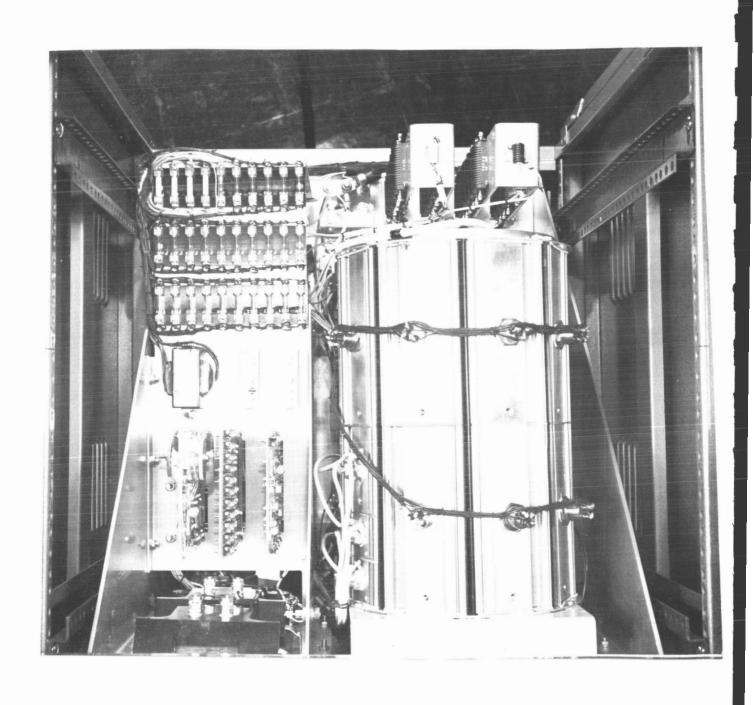


FIGURE 14. Top View of Simulator With Access Door Open

this breakpoint, and would dictate what action is required to improve the closed-loop response of the simulator. The analysis should include the effect of wiring capacitance, especially in light of the present wiring arrangement, where the collector and base leads to the power transistors run parallel. This arrangement appears similar to the well known Miller-effect in vacuum tubes, where the effect of the plate-to-grid capacitance is multiplied by the gain of the device.

In any case, of course, the need for improvements must be considered in light of specific requirements.

CONCLUSIONS

The principal objectives of this project were to develop a d-c power supply capable of simulating both the static and the dynamic characteristics of a solar-cell array. A major design problem was to obtain the transient response capability typical of a solar cell. To obtain a response of this magnitude required a very careful selection and placement of components to (a) make the system least susceptible to noise, and (b) to reduce the effects of stray capacitance.

Although slight improvements in such areas as frequency response could be obtained as discussed previously, it is believed that the simulation of solar-cell array characteristics to the extent accomplished by this simulator represents a state-of-the-art demonstration of power supply capabilities not previously attempted.

PRECEDING PAGE BLANK NOT FILMED.

APPENDIX

PARTS LIST

Reference Designation	Description
Al, A2	Operational Amplifier
C1A, C1B C2, C19 C3A, C3B, C8 C4, C5, C16 C6, C7 C12 C9 C10, C11 C13, C14, C15 C17 C18	Capacitors 100 WVDC 2250 MFD 47 MFD 50V 100 MFD 25V 1 MFD 35V 1 MFD 150V 1.5 MFD 50V 1 MFD 220V 2 MFD 220V 0.01 MFD 200 PF 0.0050 MFD
CB1	Circuit Breaker - 90A
CR1, CR2, CR3, CR4, CR5, CR6 CR7, CR8, CR14, CR16, CR21-28, CR32-35, CR39-42 CR9 CR10, CR15 CR11, CR13 CR30, CR38 CR29 CR37	1N1184 1N645 1N3026B 1N3024B 1N1200 8.2V Reference 4.7V Reference 1N2982B
F1-F3 F4 F5 F101-F125	Fuses 20 amp 1 amp 0.125 amp 5 amp
I1-I4 I5-I6	Lamp 110V Lamp 60V
Jl	Connector
Ll	Filter Choke

Reference Designation M1 M2	Description Meters 0-80V 0-100V
Pl	Connector
Q1, Q101-Q125 Q2 Q4, Q6-Q11 Q5 Q3	Transistors 2N2758 2N3430 2N2102 2N3494 MHT 6310
R1 R2 R3 R40, R41 R5 R7, R27, R29, R54, R64 R8, R39 R9 R10 R11, R20, R21 R12, R22 R13, R23 R14, R24 R15, R25, R44 R16 R17, R33 R34 R19 R26 R28, R62 R30 R38 R31 R35 R36 R37 R60 R42, R52 R45	Resistors 10K Variable 200K 1/2W CC 12 1W CC 10K 1/2W CC 51K 1/2W CC 5.1K 1/2W CC 5.1K 1/2W CC 5.1K 1/2W CC 10K Variable 1 MEG Variable 250K Variable 250K Variable 50K Variable 1 MEG 1/2W CC 39K 1/2W CC 39K 1/2W CC 39K 1/2W CC 230 OHM 20W WW 100 1/2W CC 510 5W WW
R46 R47, R49, R43, R59 R48 R50 R51	200 1W CC 2K 1/2W CC 1K 10W WW 200 1/2W CC 91 1W CC

Reference Designation	Description
R53, R66-R70 R55 R56 R57 R65 R61 R6 R4 R63 R32 R71	Resistors (Cont.) 100 50W WW 10 1/2W CC 30K 1/2W CC 510K 1/2W CC 50K Variable 18K 1/2W CC 1K 1/2W CC 7.5K 1/2W CC 3.9K 1/2W CC 820 1/2W CC 20 OHMS 1W CC 47 OHMS 1W CC
R99	Non-Inductive Shunt 100A 300 MV
R100	Shunt 100A 50 MV
R101-R125	Resistor 3 OHMS 50W WW
S1 S2	Switches 4PST 4PDT
S3,S4	Thermostat 90° C
S5	Switch SPST
T1 T2 T3 T4 T5	Transformers Main Power Auxiliary Power Magnetic Oscillator Current Sensing Voltage Sensing
B1, B2 B3	Blowers Heat Sink Transformer

DISTRIBUTION LIST FOR SUMMARY REPORT

Contract NAS3-7920

National Aeronautics and Space Administration
Washington, D. C. 20546
Attn: RNT/James Lazar (1)
RNW/Preston T. Maxwell (1)

NASA-Lewis Research Center
21000 Brookpark Road
Cleveland, Ohio 44135
Attn: B. L. Sater M.S. 54-3 (4)
J. R. Eski M.S. 54-2 (1)
R. D. Shattuck M.S. 54-3 (1)

Jesse Hall M.S. 77-1 (1) C. S. Corcoran, Jr. M.S. 500-201 (1)

F. Gourash M.S. 500-201 (1) V. R. Lalli M.S. 500-203 (1)

J. J. Weber M.S. 3-16 (1)
C. C. Conger M.S. 54-1 (1)

C. W. Knoop M.S. 54-3 (1)

Library M.S. 60-3 (2)

Reports Control Office (M.S. 5-5) (1)

Office of Reliability and Quality Assurance M.S. 500-203 (1)

NASA Scientific and Technical Information Facility (6) P. O. Box 33 College Park, Maryland 20740 Attn: NASA Representative ROT-2448

NASA Ames Research Center (1) Moffett Field, California 94035 Attn: Library

NASA Flight Research Center (1)
P. O. Box 273
Edwards, California 93423
Attn: Library

NASA Goddard Space Flight Center Greenbelt, Maryland 20771 Attn: Library (1) Robert L. Weitzel (1) Fred C. Yagerhofer (1) Jet Propulsion Laboratory
4800 Oak Grove Drive
Pasadena, California 91103
Attn: Library (1)
Jack Sterns (1)

NASA Langley Research Center (1) Langley Station Hampton, Virginia 23365 Attn: Library

NASA Manned Spacecraft Center (1) Houston, Texas 77001 Attn: Library

NASA Marshall Space Flight Center Huntsville, Alabama 35812 Attn: Library (1) Robert Harwell (1) Joseph Parker (1)

NASA Western Operations (1) 150 Pico Boulevard Santa Monica, California 90406 Attn: Library

AFWL (1)
Kirtland Air Force Base, New Mexico 87417
Attn: Capt. C. F. Ellis/WLPC

Aerospace Corporation (1)
P. O. Box 95085
Los Angeles, California 90045
Attn: Library Technical Documents Group

Westinghouse Astronuclear Laboratories (1)
Pittsburgh, Pennsylvania 15234
Attn: Library

Westinghouse Electric Corporation
Aerospace Electrical Division
P. O. Box 989
Lima, Ohio 45802
Attn: Library (1)
H. B. James (1)

Westinghouse Electric Corporation (1)
P. O. Box 10596
Pittsburgh, Pennsylvania 15235
Attn: A. H. B. Walker

Electro-Optical Systems, Inc. (1) 125 N. Vinedo Avenue Pasadena, California 91107 Attn: John Davis

General Electric Company (1) Speciality Control Department Waynesboro, Virginia 22980 Attn: Army L. Wilford

TRW Inc. (1)
Electromechanical Division
23555 Euclid Avenue
Cleveland, Ohio 44117
Attn: Paul H. McGarrell

Hughes Research Laboratories (1) 3011 Malibu Canyon Road Malibu, California 93032 Attn: Dr. George R. Brewer

TRW Inc. (1)
TRW Systems Group
One Space Park
Redondo Beach, California 90278
Attn: Dr. F. Kraus

Hughes Aircraft Company (1) Century Blvd. Los Angeles, California Attn: W. J. Muldoon

NASA-Electronics Research Center 575 Technology Square Cambridge, Massachusetts 02039 Attn: Library (1) Dr. Frederick Niemann (1) Duke University
Durham, North Carolina 27706
Attn: Dean, College of Engineering (1)
Dr. Thomas G. Wilson, Chairman E.E. Dept. (1)

U. S. Army Electronics Laboratory (1)
Power Conditioning Sion
Et. Monmouth, New Jersey
Attn: Mr. Frank Wrublewski